

## WAVE PROPAGATION THROUGH A NOZZLE WITH ELASTIC WALLS

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*Se studiază propagarea micilor perturbații pentru o problemă simplă de curgere-structură. Este considerată curgerea unui fluid izentropic, compresibil, nevâscos printr-o doză cu pereți elastici. În prezența structurii elementului frontieră al fluidului cercetăm influența numărului Mach al mișcării neperturbate asupra vitezei de propagare a undelor.*

*Study of small perturbations propagation in a simple flow-structure problem shall be made. The flow of a compressible inviscid and isentropic fluid through a nozzle with elastic walls is presented. In presence of a coupling with a structural element bounding the fluid we investigate the influence of Mach number of the unperturbed flow on the speed of propagating waves.*

**Keywords:** Wave propagation, small perturbations, elastic nozzle walls.

**Mathematics Subject Classification 2000:** 76B15, 74B15, 35Q35.

### Introduction

We study the propagation of small perturbations in a nozzle with parallel elastic walls (see [1], [2]). We consider a bi-dimensional inviscid, isentropic and compressible fluid flow through a nozzle with elastic walls. For initial time we suppose that the nozzle has straight walls. The study is divided in two parts. We study in first section the one-dimensional flow and in second the two-dimensional flow-structure interaction.

#### 1. One-dimensional flow-structure problem

Denoting  $c(x, t)$ , the local speed of sound,  $u(x, t)$  the fluid velocity in  $x$  direction and  $H(x, t)$  the nozzle height,  $u_0(x, t)$  shall be the initial fluid velocity.

The lateral section of the nozzle is illustrated in Fig 1.

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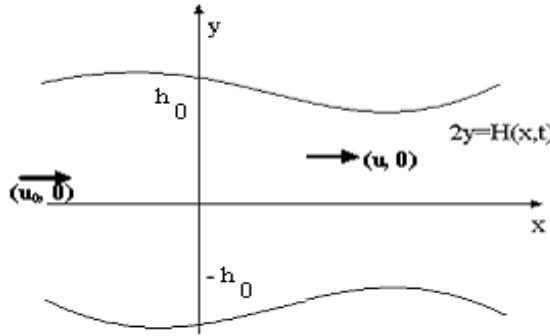


Fig 1. Lateral section of the nozzle.

Under the hypothesis that the walls of the nozzle are so thin that the motion is governed by the linear beam equation (see [1]) the equations governing the flow are:

$$\begin{aligned} \frac{2 \cdot c_t}{\gamma - 1} + c u_x + \frac{2 \cdot c_x u_0}{\gamma - 1} + \frac{c}{H} (H_t + u_0 H_x) &= 0, \quad u_t + u_0 u_x + \frac{2}{\gamma - 1} c_0 c_x = 0, \\ H_{tt} + D H_{xxxx} &= m(p_i - p_0), \end{aligned} \quad (1.1)$$

for  $\gamma$  the specific heats ratio,  $D$  the bending stiffness,  $p_i$  the local pressure of the fluid,  $p_0$  the outside ambient pressure and  $m$  the linear mass of the walls that shall be supposed unity.

The evolution of small perturbations for the system (1.1) is expressed by functions  $c', u', H'$  for  $c = c_0 + c'$ ,  $u = u_0 + u'$ ,  $H = H_0 + H'$ . The system for perturbations shall be obtained if we assume that prime quantities are small comparing with those in the unperturbed flow denoted with  $c_0, u_0, H_0$  for  $p_0 = 1$ ,  $\rho_0 = 1$ . Dropping the prime notation for perturbations and using known relations:  $p = p_0 (1 + (\gamma - 1) M_0^2 / 2)^{-\gamma / (\gamma - 1)}$ ,  $\rho = \rho_0 (1 + (\gamma - 1) M_0^2 / 2)^{-1 / (\gamma - 1)}$ ,  $c_0^2 / (\gamma - 1) = c^2 / (\gamma - 1) + u^2 / 2$  we have for first approximation:  $p - p_0 = 2c_0 c / (\gamma - 1) \cdot (c_0^2 / \gamma)^{1 / (\gamma - 1)}$  and the system of equations for perturbations is:

$$\begin{aligned} 2c_t / (\gamma - 1) + c_0 u_x + 2c_x u_0 / (\gamma - 1) + c_0 (H_t + u_0 H_x) / H_0 &= 0, \\ (\gamma - 1)(u_t + u_0 u_x) + 2c_0 c_x &= 0, \\ H_{tt} + D H_{xxxx} - 2 \cdot c_0 (c_0^2 / \gamma)^{1 / (\gamma - 1)} c / (\gamma - 1) &= 0. \end{aligned} \quad (1.2)$$

We search for solutions through simple waves such that for  $k, \omega \neq 0$

$$c(x, t) = \varphi(kx - \omega t), \quad u(x, t) = \psi(kx - \omega t), \quad H(x, t) = h(kx - \omega t), \quad (1.3)$$

for  $\varphi, \psi \in C^{(1)}(\mathbf{R})$ ,  $h \in C^{(4)}(\mathbf{R})$ . From (1.2) with  $A_0 = (c_0^2 / \gamma)^{1/(\gamma-1)}$  we find:

$$\begin{aligned} 2(ku_0 - \omega)\varphi' / (\gamma - 1) + c_0 k \psi' + c_0 (ku_0 - \omega)h' / H_0 &= 0, \\ (ku_0 - \omega)\psi' + 2c_0 k \varphi' / (\gamma - 1) &= 0. \end{aligned} \quad (1.4)$$

If  $ku_0 - \omega = 0$  then  $\varphi = ct.$ ,  $\psi = ct.$ ,  $h = ct.$  expressing a permanent flow.

We shall continue under the case  $ku_0 - \omega \neq 0$  meaning  $\frac{\omega}{k} \neq u_0$  in which phase velocity differs from the flow speed. We can write from (1.4):

$$\begin{aligned} \psi' &= -\frac{c_0 k}{(\gamma - 1)(ku_0 - \omega)} \varphi', \quad h' = \frac{(ku_0 - \omega)^2}{c_0^2 k^2} \varphi', \\ Dk^4 \frac{(ku_0 - \omega)^2}{c_0^2 k^2} \varphi^{(4)} - \frac{\omega^2 (ku_0 - \omega)^2}{c_0^2 k^2} \varphi'' - \frac{2A_0}{\gamma - 1} c_0 \varphi &= 0 \end{aligned} \quad (1.5)$$

Solving differential equation (1.5)<sub>3</sub> and denoting  
 $\lambda = \frac{1}{k^2 \sqrt{2D}} \sqrt{\omega^2 + \frac{\sqrt{\delta}}{|ku_0 - \omega|}}$ ,  $\delta = \omega^4 + \frac{8Dk^6 A_0 c_0^3}{\gamma - 1} > 0$  we shall find the fundamental solutions:

$$\varphi_1 = \sin \lambda \xi, \quad \varphi_2 = \cos \lambda \xi, \quad \varphi_3 = \sin \alpha \xi, \quad \varphi_4 = \cos \alpha \xi; \quad (1.6)$$

$$\text{for } \omega^2 > \frac{\sqrt{\delta}}{|ku_0 - \omega|}, \quad \alpha = \frac{1}{k^2 \sqrt{2D}} \sqrt{\omega^2 - \frac{\sqrt{\delta}}{|ku_0 - \omega|}}, \quad (1.7)$$

$$\varphi_1 = \sin \lambda \xi, \quad \varphi_2 = \cos \lambda \xi, \quad \varphi_3 = e^{\alpha \xi}, \quad \varphi_4 = e^{-\alpha \xi}; \quad (1.8)$$

$$\text{for } \omega^2 < \frac{\sqrt{\delta}}{|ku_0 - \omega|}, \quad \alpha = \frac{1}{k^2 \sqrt{2D}} \sqrt{\frac{\sqrt{\delta}}{|ku_0 - \omega|} - \omega^2}, \quad (1.9)$$

$$\text{or } \varphi_1 = \sin \lambda \xi, \quad \varphi_2 = \cos \lambda \xi, \quad \varphi_3 = A \xi + B, \quad \text{for } \omega^2 = \frac{\sqrt{\delta}}{|ku_0 - \omega|}. \quad (1.10)$$

In order to investigate the influence of Mach number of the unperturbed flow on the speed of propagating waves we shall make graphical representations

of level curves for the function  $f(\omega, k) = \omega^2 - \frac{\sqrt{\delta}}{|ku_0 - \omega|}$ . We shall use the following constants:  $c_0 = \sqrt{1.4}$ ,  $\gamma = 1.4025$ ,  $D$  of order  $10^{-3}$ , Mach number  $M_0 = u_0 / c_0 \in [0, 13]$  and  $\omega/k \in [-1, 1]$ . Representations are made in Figs 2. and 3.

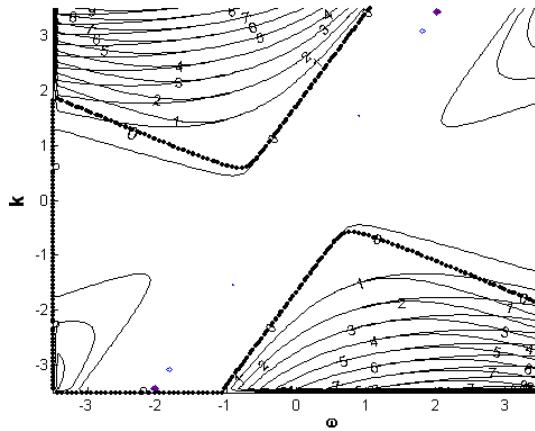
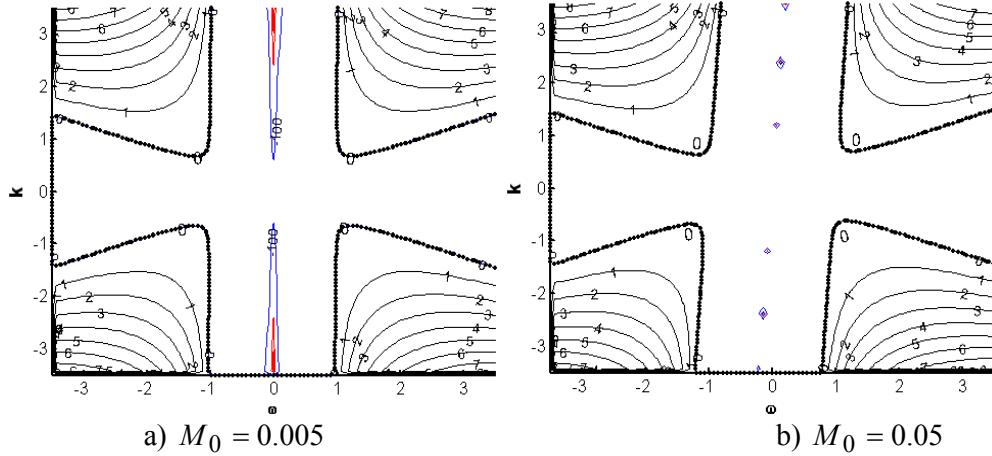
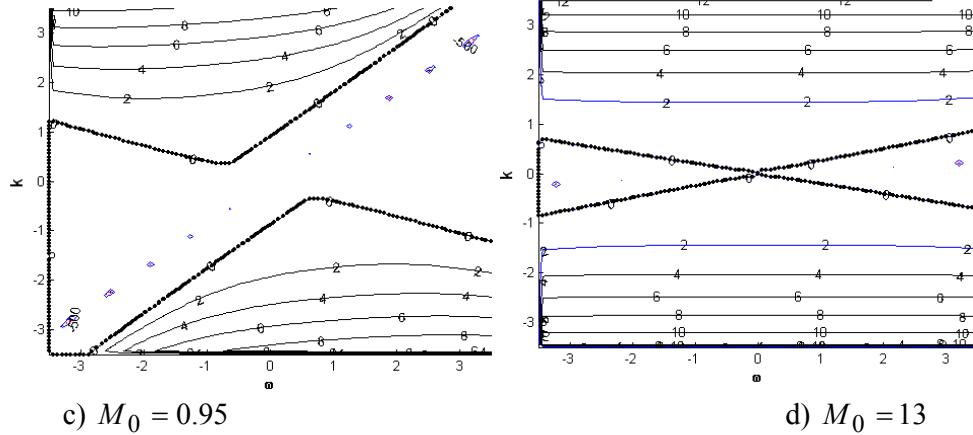


Fig. 2. Variation of  $D$   $D=0.001$ ,  $D=0.005$  for fixed  $M_0 = 0.5$



Fig. 3. Variation of Mach number for fixed  $D=0.001$ .

From Figs 2. and 3. we conclude that the domain  $\omega/k < 0$  reduces once Mach number is increasing.

For  $\xi = kx - \omega t$  we found bounded general solution for (1.3)<sub>1</sub>:

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t) + F_3 \sin \alpha(kx - \omega t) + F_4 \cos \alpha(kx - \omega t), \quad (1.11)$$

for  $\omega^2 > \frac{\sqrt{\delta}}{|ku_0 - \omega|}$ , and in case  $\omega^2 < \frac{\sqrt{\delta}}{|ku_0 - \omega|}$  the local speed of sound becomes:

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t), \quad \frac{\omega}{k} > 0,$$

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t) + F_3 e^{\alpha(-kx + \omega t)}, \quad k > 0 > \omega \quad (1.12)$$

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t) + F_3 e^{\alpha(kx - \omega t)}, \quad k < 0 < \omega.$$

Then from (1.5) after integration we can write:

$$H(x, t) = \frac{(ku_0 - \omega)^2}{c_0^2 k^2} c(x, t) + C_1; \quad u(x, t) = -\frac{c_0 k}{(\gamma - 1)(ku_0 - \omega)} c(x, t) + C_2. \quad (1.13)$$

Using initial conditions:  $c(x, t) = c_0 C(x)$ ,  $u(x, t) = u_0 U(x)$ ,  $H(x, t) = H_0 f(x)$ , and boundary conditions we can express the constants of integration. Also from these conditions one obtains a *compatibility relation between initial conditions* for

existence of motion through simple waves:  $\frac{(ku_0 - \omega)^2}{c_0^2 k^2} c_0 C(x) + C_1 = H_0 f(x)$ ,

$$-\frac{c_0 k}{(\gamma - 1)(ku_0 - \omega)} c_0 C(x) + C_2 = u_0 U(x), \quad \text{from where:}$$

$$u_0 U(x) - C_2 = c_0^3 k^3 / (\gamma - 1) (k u_0 - \omega)^3 (C_1 - H_0 f(x)). \quad (1.14)$$

**Remarks:**

1. If  $U(x) = ct.$  or  $f(x) = ct.$  then  $f(x) = ct., C(x) = ct.$  or  $U(x) = ct., C(x) = ct.$

Looking for solution which has:  $c(x,t) = c_0 \varepsilon, u(x,t) = 0, H(x,t) = H_0 \varepsilon$  or  $c(x,t) = c_0 \varepsilon, u(x,t) = u_0 \varepsilon, H(x,t) = 0, \varepsilon = O(10^{-2})$  we find

$$F_1 \sin \lambda kx + F_2 \cos \lambda kx + F_3 \sin \alpha kx + F_4 \cos \alpha kx = c_0 C,$$

$$\frac{(k u_0 - \omega)^2}{c_0^2 k^2} c_0 C + C_1 = H_0 \varepsilon, \frac{c_0 k}{(\gamma - 1)(k u_0 - \omega)} c_0 C + C_2 = u_0 \varepsilon.$$

2. *Boundary conditions* could be imposed if we consider the domain  $x \in [0, L], y \in [0, H(x)]$  at each  $t: H(0,t) = H(L,t) = 0, H_{xx}(0,t) = H_{xx}(L,t) = 0.$

Considering also the case  $f(x) \neq ct., U(x) \neq ct.$  we look for the constants  $F_i, i=1,2,3,4$  in order to obtain the general solution.

## 2. Bi-dimensional flow-structure problem

The fluid velocity has now two components, on  $x$  and  $y$  direction. The fluid flow is sketched in Fig. 4.

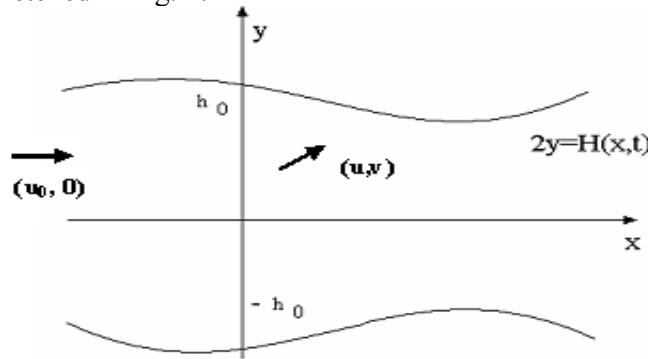


Fig. 4. Lateral section of the nozzle.

We consider a potential flow, with the potential  $\phi(x, y, t)$  and flow velocity vector  $\vec{V} = u_0 \vec{i} + \vec{v}, \vec{v} = (u, v) = (\phi_x, \phi_y).$  The equations governing the flow are:

$$(1 - M_0^2) \phi_{xx} + \phi_{yy} - (2u_0 \phi_{xt} + \phi_{tt}) / c_0^2 = 0,$$

$$H_{tt} + DH_{xxxx} = p_i - p_0 = 2\rho_0 (\phi_t + u_0 \phi_x) \text{ on } 2y = H(x, t), \quad (2.1)$$

where  $(u_0, 0)$  is the velocity vector in the unperturbed flow and  $\phi, H$  bounded functions for  $y \rightarrow \pm\infty$  (see [3], [4], [5]).

**Remarks:**

1. For bi-dimensional inviscid, isentropic and compressible fluid the pressure and mass density could be written as:  $p = p_0(1 + (\gamma - 1)(\phi_t + V^2/2)/c_0^2)^{\gamma/(\gamma-1)}$ ,  $\rho = \rho_0(1 + (\gamma - 1)(\phi_t + V^2/2)/c_0^2)^{1/(\gamma-1)}$  with the module of velocity:  $V^2 = (u_0 + u)^2 + v^2$ . Then for small perturbations  $V^2 = u_0^2 + 2u_0u$  and pressure on  $2y = H(x, t)$  becomes:  $p = p_0(1 - \gamma(\phi_t + \frac{1}{2}2u_0u)/c_0^2 + \dots)$  from where  $p_0 - p = \rho_0(\phi_t + u_0\phi_x)$ .

2. Boundary condition on the surface  $2y = H(x, t)$  must be imposed for potential obtained from velocity in x direction:  $\phi_y = H_t + u_0 H_x$ .

For function H we shall consider initial and boundary conditions:

$H(x, 0) = H_0 f(x)$ , and  $H(0, t) = H(L, t) = 0$ ,  $H_{xx}(0, t) = H_{xx}(L, t) = 0$ .

Looking for motion **through simple waves** we consider:

$\phi(x, y, t) = F(k_1 x + k_2 y - \omega t)$ ,  $H(x, t) = h(k_1 x - \omega t)$ ,  $F \in C^{(2)}(\mathbf{R})$ ,  $h \in C^{(4)}(\mathbf{R})$ .

From (2.1) we find:

$$(k_1^2 + k_2^2 - (\omega/c_0 - M_0 k_1)^2)F'' = 0, \\ \omega^2 h'' + Dk_1^4 h^{(4)} = 2\rho_0(k_1 u_0 - \omega)F'. \quad (2.2)$$

For  $F'' \neq 0$  one find the **dispersion equation**:

$$k_2 = \pm \sqrt{\left(\frac{\omega}{c_0} - M_0 k_1\right)^2 - k_1^2} = \pm \sqrt{\frac{(\omega - k_1 u_0)^2}{c_0^2} - k_1^2}.$$

In order to solve equation (2.1)<sub>1</sub> we change the variables  $(x, t)$  through  $(\xi, \eta)$  with:  $\xi = x - (u_0 + c_0)t$ ,  $\eta = x - (u_0 - c_0)t$  obtaining a new equation:

$$4\phi_{\xi\eta} + \phi_{yy} = 0, \quad \phi = \phi(\xi, \eta, y). \quad (2.3)$$

that will be solved considering a separation of variables  $(\xi, \eta)$  from y variables:

$\phi = \phi(\xi, \eta)F(y)$ . We find  $\frac{\phi_{\xi\eta}}{\phi} = -\frac{1}{4} \frac{F''(y)}{F(y)} = K$ ,  $K > 0$  for bounded solutions on y.

Then  $F(y) = A \sin 2\lambda y + B \cos 2\lambda y$  and  $\phi_{\xi\eta} - \lambda^2 \phi = 0$ . Solution obtained for  $\lambda = 0$  is  $\phi(\xi, \eta) = C\xi + D\eta$  and for  $\lambda \neq 0$  is  $\phi(\xi, \eta) = Ce^{\lambda(\xi+\eta)} + De^{-\lambda(\xi+\eta)}$ . We can write the solution of (2.1)<sub>1</sub>:

$$\phi(x, y, t) = [Ce^{2\lambda(x-u_0t)} + De^{-2\lambda(x-u_0t)}][A \sin 2\lambda y + B \cos 2\lambda y]. \quad (2.4)$$

We remark that for solution (2.4) of (2.1)<sub>1</sub> we have  $\phi_t + u_0\phi_x = 0$ .

Solving equation (2.1)<sub>2</sub> we obtain for function  $H(x, t)$  the problem:

$$\begin{aligned} H_{tt} + DH_{xxxx} &= 0, \quad D = [0, L] \times (0, \infty), \\ H(x, 0) &= H_0 f(x), \\ H(0, t) &= H(L, t) = 0, \quad H_{xx}(0, t) = H_{xx}(L, t) = 0. \end{aligned} \quad (2.5)$$

Considering a separation of variables  $H(x, t) = X(x)T(t) \Rightarrow \frac{T''}{T} = -D \frac{X^{(4)}}{X} = \alpha$

and for  $\alpha > 0$  we can write  $T'' - \alpha T = 0$ ,  $X^{(4)} + (\alpha/D)X = 0$ .

With  $\beta = \sqrt[4]{\alpha/D}/\sqrt{2}$ , general solution for (2.5) is:

$$\begin{aligned} X(x) &= \cos \beta x \cdot (C_1 e^{\beta x} + C_3 e^{-\beta x}) + \sin \beta x \cdot (C_2 e^{\beta x} + C_4 e^{-\beta x}), \\ T(t) &= Pe^{\sqrt{\alpha}t} + Qe^{-\sqrt{\alpha}t}, \quad P = 0, \text{ for a bounded solution on t.} \end{aligned} \quad (2.6)$$

Solution for equation (2.5)<sub>1</sub> is:

$$H(x, t) = \{\cos \beta x [C_1 e^{\beta x} + C_3 e^{-\beta x}] + \sin \beta x [C_2 e^{\beta x} - C_4 e^{-\beta x}]\} Q e^{-\sqrt{\alpha}t} \quad (2.7)$$

From conditions (2.5)<sub>2</sub> we find:

$$\begin{aligned} H(0, t) = 0 &\Rightarrow C_3 = -C_1, \quad H_{xx}(0, t) = 0 \Rightarrow C_2 = C_4, \\ H(L, t) = 0 &\Rightarrow C_1 \cos \beta L (e^{\beta L} - e^{-\beta L}) + \sin \beta L (C_2 e^{\beta L} + C_4 e^{-\beta L}) = 0, \\ H_{xx}(L, t) = 0 &\Rightarrow 2\beta L = -\frac{\pi}{2} + 2k\pi, \quad k \in N \Rightarrow \sqrt{\alpha} = \sqrt{2D}(-\pi/4 + k\pi)^2/L^2, \end{aligned} \quad (2.8)$$

and (2.7)  $\Rightarrow C_2 = C_1 t h(-\pi/4 + k\pi)$  for each  $k$ .

The general solution for (2.5) is:

$$H(x,t) = \sum_{k=1}^{\infty} M_k \{ \cos((-\pi/4 + k\pi)x/L) sh((-\pi/4 + k\pi)x/L) + \\ + t h(-\pi/4 + k\pi) \sin((-\pi/4 + k\pi)x/L) ch((-\pi/4 + k\pi)x/L) \} e^{-\frac{\sqrt{2D}}{L^2} (-\frac{\pi}{4} + k\pi)^2 t} \quad (2.9)$$

with  $M_k$  determined from  $H(x,0) = H_0 f(x)$  for which:

$$H_0 f(x) = \sum_{k=1}^{\infty} M_k \{ \cos((-\pi/4 + k\pi)x/L) sh((-\pi/4 + k\pi)x/L) + \\ + t h(-\pi/4 + k\pi) \sin((-\pi/4 + k\pi)x/L) ch((-\pi/4 + k\pi)x/L) \}. \quad (2.10)$$

### Conclusions

For one-dimensional case was investigated the influence of Mach number of the unperturbed flow upon the speed of propagating waves (presented in Figs 2 and 3).

For the bi-dimensional case instead of a discussion of the dispersion equation we have studied the solution for potential using condition on the boundary:  $\phi_y = H_t + u_0 H_x$  on  $2y = H(x,t)$ , and initial conditions on velocity. The velocity field is expressed by:

$$u = \phi_x = 2\lambda [C e^{2\lambda(x-u_0 t)} - D e^{-2\lambda(x-u_0 t)}] [A \sin 2\lambda y + B \cos 2\lambda y], \\ v = \phi_y = 2\lambda [C e^{2\lambda(x-u_0 t)} + D e^{-2\lambda(x-u_0 t)}] [A \cos 2\lambda y - B \sin 2\lambda y].$$

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