

WAVE PROPAGATION THROUGH A NOZZLE WITH ELASTIC WALLS

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Se studiază propagarea micilor perturbații pentru o problemă simplă de curgere-structură. Este considerată curgerea unui fluid izentropic, compresibil, nevâscos printr-o doză cu pereți elastici. În prezența structurii elementului frontieră al fluidului cercetăm influența numărului Mach al mișcării neperturbate asupra vitezei de propagare a undelor.

Study of small perturbations propagation in a simple flow-structure problem shall be made. The flow of a compressible inviscid and isentropic fluid through a nozzle with elastic walls is presented. In presence of a coupling with a structural element bounding the fluid we investigate the influence of Mach number of the unperturbed flow on the speed of propagating waves.

Keywords: Wave propagation, small perturbations, elastic nozzle walls.

Mathematics Subject Classification 2000: 76B15, 74B15, 35Q35.

Introduction

We study the propagation of small perturbations in a nozzle with parallel elastic walls (see [1], [2]). We consider a bi-dimensional inviscid, isentropic and compressible fluid flow through a nozzle with elastic walls. For initial time we suppose that the nozzle has straight walls. The study is divided in two parts. We study in first section the one-dimensional flow and in second the two-dimensional flow-structure interaction.

1. One-dimensional flow-structure problem

Denoting $c(x, t)$, the local speed of sound, $u(x, t)$ the fluid velocity in x direction and $H(x, t)$ the nozzle height, $u_0(x, t)$ shall be the initial fluid velocity.

The lateral section of the nozzle is illustrated in Fig 1.

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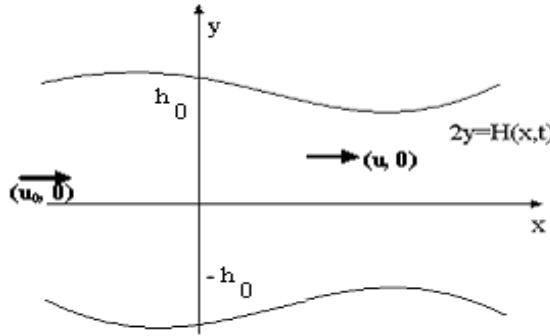


Fig 1. Lateral section of the nozzle.

Under the hypothesis that the walls of the nozzle are so thin that the motion is governed by the linear beam equation (see [1]) the equations governing the flow are:

$$\begin{aligned} \frac{2 \cdot c_t}{\gamma - 1} + c u_x + \frac{2 \cdot c_x u_0}{\gamma - 1} + \frac{c}{H} (H_t + u_0 H_x) &= 0, \quad u_t + u_0 u_x + \frac{2}{\gamma - 1} c_0 c_x = 0, \\ H_{tt} + D H_{xxxx} &= m(p_i - p_0), \end{aligned} \quad (1.1)$$

for γ the specific heats ratio, D the bending stiffness, p_i the local pressure of the fluid, p_0 the outside ambient pressure and m the linear mass of the walls that shall be supposed unity.

The evolution of small perturbations for the system (1.1) is expressed by functions c', u', H' for $c = c_0 + c'$, $u = u_0 + u'$, $H = H_0 + H'$. The system for perturbations shall be obtained if we assume that prime quantities are small comparing with those in the unperturbed flow denoted with c_0, u_0, H_0 for $p_0 = 1$, $\rho_0 = 1$. Dropping the prime notation for perturbations and using known relations: $p = p_0 (1 + (\gamma - 1) M_0^2 / 2)^{-\gamma / (\gamma - 1)}$, $\rho = \rho_0 (1 + (\gamma - 1) M_0^2 / 2)^{-1 / (\gamma - 1)}$, $c_0^2 / (\gamma - 1) = c^2 / (\gamma - 1) + u^2 / 2$ we have for first approximation: $p - p_0 = 2c_0 c / (\gamma - 1) \cdot (c_0^2 / \gamma)^{1 / (\gamma - 1)}$ and the system of equations for perturbations is:

$$\begin{aligned} 2c_t / (\gamma - 1) + c_0 u_x + 2c_x u_0 / (\gamma - 1) + c_0 (H_t + u_0 H_x) / H_0 &= 0, \\ (\gamma - 1)(u_t + u_0 u_x) + 2c_0 c_x &= 0, \\ H_{tt} + D H_{xxxx} - 2 \cdot c_0 (c_0^2 / \gamma)^{1 / (\gamma - 1)} c / (\gamma - 1) &= 0. \end{aligned} \quad (1.2)$$

We search for solutions through simple waves such that for $k, \omega \neq 0$

$$c(x, t) = \varphi(kx - \omega t), \quad u(x, t) = \psi(kx - \omega t), \quad H(x, t) = h(kx - \omega t), \quad (1.3)$$

for $\varphi, \psi \in C^{(1)}(\mathbf{R})$, $h \in C^{(4)}(\mathbf{R})$. From (1.2) with $A_0 = (c_0^2 / \gamma)^{1/(\gamma-1)}$ we find:

$$\begin{aligned} 2(ku_0 - \omega)\varphi' / (\gamma - 1) + c_0 k \psi' + c_0 (ku_0 - \omega)h' / H_0 &= 0, \\ (ku_0 - \omega)\psi' + 2c_0 k \varphi' / (\gamma - 1) &= 0. \end{aligned} \quad (1.4)$$

If $ku_0 - \omega = 0$ then $\varphi = ct.$, $\psi = ct.$, $h = ct.$ expressing a permanent flow.

We shall continue under the case $ku_0 - \omega \neq 0$ meaning $\frac{\omega}{k} \neq u_0$ in which phase velocity differs from the flow speed. We can write from (1.4):

$$\begin{aligned} \psi' &= -\frac{c_0 k}{(\gamma - 1)(ku_0 - \omega)} \varphi', \quad h' = \frac{(ku_0 - \omega)^2}{c_0^2 k^2} \varphi', \\ Dk^4 \frac{(ku_0 - \omega)^2}{c_0^2 k^2} \varphi^{(4)} - \frac{\omega^2 (ku_0 - \omega)^2}{c_0^2 k^2} \varphi'' - \frac{2A_0}{\gamma - 1} c_0 \varphi &= 0 \end{aligned} \quad (1.5)$$

Solving differential equation (1.5)₃ and denoting
 $\lambda = \frac{1}{k^2 \sqrt{2D}} \sqrt{\omega^2 + \frac{\sqrt{\delta}}{|ku_0 - \omega|}}$, $\delta = \omega^4 + \frac{8Dk^6 A_0 c_0^3}{\gamma - 1} > 0$ we shall find the fundamental solutions:

$$\varphi_1 = \sin \lambda \xi, \quad \varphi_2 = \cos \lambda \xi, \quad \varphi_3 = \sin \alpha \xi, \quad \varphi_4 = \cos \alpha \xi; \quad (1.6)$$

$$\text{for } \omega^2 > \frac{\sqrt{\delta}}{|ku_0 - \omega|}, \quad \alpha = \frac{1}{k^2 \sqrt{2D}} \sqrt{\omega^2 - \frac{\sqrt{\delta}}{|ku_0 - \omega|}}, \quad (1.7)$$

$$\varphi_1 = \sin \lambda \xi, \quad \varphi_2 = \cos \lambda \xi, \quad \varphi_3 = e^{\alpha \xi}, \quad \varphi_4 = e^{-\alpha \xi}; \quad (1.8)$$

$$\text{for } \omega^2 < \frac{\sqrt{\delta}}{|ku_0 - \omega|}, \quad \alpha = \frac{1}{k^2 \sqrt{2D}} \sqrt{\frac{\sqrt{\delta}}{|ku_0 - \omega|} - \omega^2}, \quad (1.9)$$

$$\text{or } \varphi_1 = \sin \lambda \xi, \quad \varphi_2 = \cos \lambda \xi, \quad \varphi_3 = A \xi + B, \quad \text{for } \omega^2 = \frac{\sqrt{\delta}}{|ku_0 - \omega|}. \quad (1.10)$$

In order to investigate the influence of Mach number of the unperturbed flow on the speed of propagating waves we shall make graphical representations

of level curves for the function $f(\omega, k) = \omega^2 - \frac{\sqrt{\delta}}{|ku_0 - \omega|}$. We shall use the following constants: $c_0 = \sqrt{1.4}$, $\gamma = 1.4025$, D of order 10^{-3} , Mach number $M_0 = u_0 / c_0 \in [0, 13]$ and $\omega/k \in [-1, 1]$. Representations are made in Figs 2. and 3.

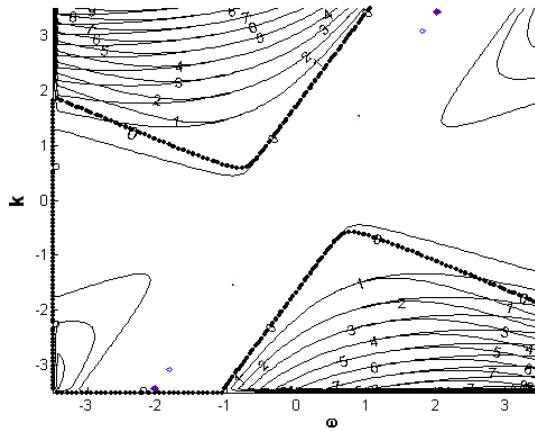
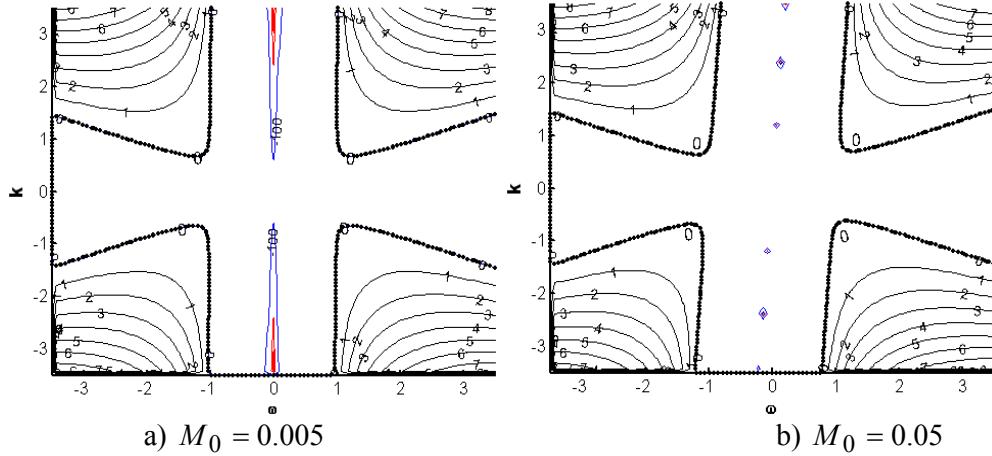
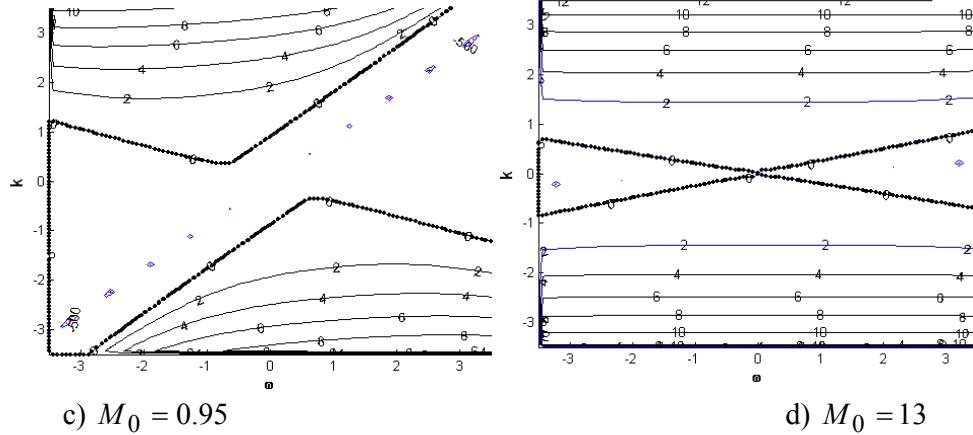


Fig. 2. Variation of D $D=0.001$, $D=0.005$ for fixed $M_0 = 0.5$



Fig. 3. Variation of Mach number for fixed $D=0.001$.

From Figs 2. and 3. we conclude that the domain $\omega/k < 0$ reduces once Mach number is increasing.

For $\xi = kx - \omega t$ we found bounded general solution for (1.3)₁:

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t) + F_3 \sin \alpha(kx - \omega t) + F_4 \cos \alpha(kx - \omega t), \quad (1.11)$$

for $\omega^2 > \frac{\sqrt{\delta}}{|ku_0 - \omega|}$, and in case $\omega^2 < \frac{\sqrt{\delta}}{|ku_0 - \omega|}$ the local speed of sound becomes:

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t), \quad \frac{\omega}{k} > 0,$$

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t) + F_3 e^{\alpha(-kx + \omega t)}, \quad k > 0 > \omega \quad (1.12)$$

$$c(x, t) = F_1 \sin \lambda(kx - \omega t) + F_2 \cos \lambda(kx - \omega t) + F_3 e^{\alpha(kx - \omega t)}, \quad k < 0 < \omega.$$

Then from (1.5) after integration we can write:

$$H(x, t) = \frac{(ku_0 - \omega)^2}{c_0^2 k^2} c(x, t) + C_1; \quad u(x, t) = -\frac{c_0 k}{(\gamma - 1)(ku_0 - \omega)} c(x, t) + C_2. \quad (1.13)$$

Using initial conditions: $c(x, t) = c_0 C(x)$, $u(x, t) = u_0 U(x)$, $H(x, t) = H_0 f(x)$, and boundary conditions we can express the constants of integration. Also from these conditions one obtains a *compatibility relation between initial conditions* for

existence of motion through simple waves: $\frac{(ku_0 - \omega)^2}{c_0^2 k^2} c_0 C(x) + C_1 = H_0 f(x)$,

$$-\frac{c_0 k}{(\gamma - 1)(ku_0 - \omega)} c_0 C(x) + C_2 = u_0 U(x), \quad \text{from where:}$$

$$u_0 U(x) - C_2 = c_0^3 k^3 / (\gamma - 1) (k u_0 - \omega)^3 (C_1 - H_0 f(x)). \quad (1.14)$$

Remarks:

1. If $U(x) = ct.$ or $f(x) = ct.$ then $f(x) = ct., C(x) = ct.$ or $U(x) = ct., C(x) = ct.$

Looking for solution which has: $c(x,t) = c_0 \varepsilon, u(x,t) = 0, H(x,t) = H_0 \varepsilon$ or $c(x,t) = c_0 \varepsilon, u(x,t) = u_0 \varepsilon, H(x,t) = 0, \varepsilon = O(10^{-2})$ we find

$$F_1 \sin \lambda kx + F_2 \cos \lambda kx + F_3 \sin \alpha kx + F_4 \cos \alpha kx = c_0 C,$$

$$\frac{(k u_0 - \omega)^2}{c_0^2 k^2} c_0 C + C_1 = H_0 \varepsilon, \frac{c_0 k}{(\gamma - 1)(k u_0 - \omega)} c_0 C + C_2 = u_0 \varepsilon.$$

2. *Boundary conditions* could be imposed if we consider the domain $x \in [0, L], y \in [0, H(x)]$ at each $t: H(0,t) = H(L,t) = 0, H_{xx}(0,t) = H_{xx}(L,t) = 0.$

Considering also the case $f(x) \neq ct., U(x) \neq ct.$ we look for the constants $F_i, i=1,2,3,4$ in order to obtain the general solution.

2. Bi-dimensional flow-structure problem

The fluid velocity has now two components, on x and y direction. The fluid flow is sketched in Fig. 4.

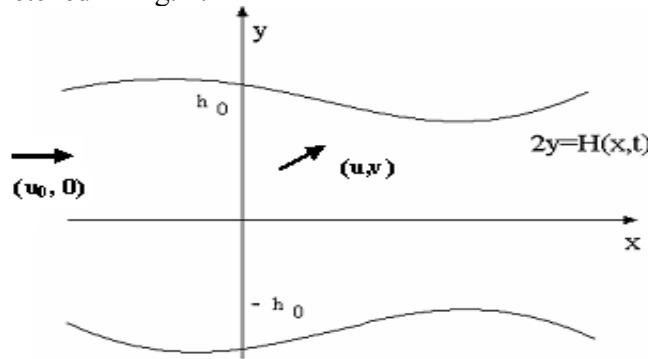


Fig. 4. Lateral section of the nozzle.

We consider a potential flow, with the potential $\phi(x, y, t)$ and flow velocity vector $\vec{V} = u_0 \vec{i} + \vec{v}, \vec{v} = (u, v) = (\phi_x, \phi_y).$ The equations governing the flow are:

$$(1 - M_0^2) \phi_{xx} + \phi_{yy} - (2u_0 \phi_{xt} + \phi_{tt}) / c_0^2 = 0,$$

$$H_{tt} + DH_{xxxx} = p_i - p_0 = 2\rho_0 (\phi_t + u_0 \phi_x) \text{ on } 2y = H(x, t), \quad (2.1)$$

where $(u_0, 0)$ is the velocity vector in the unperturbed flow and ϕ, H bounded functions for $y \rightarrow \pm\infty$ (see [3], [4], [5]).

Remarks:

1. For bi-dimensional inviscid, isentropic and compressible fluid the pressure and mass density could be written as: $p = p_0(1 + (\gamma - 1)(\phi_t + V^2/2)/c_0^2)^{\gamma/(\gamma-1)}$, $\rho = \rho_0(1 + (\gamma - 1)(\phi_t + V^2/2)/c_0^2)^{1/(\gamma-1)}$ with the module of velocity: $V^2 = (u_0 + u)^2 + v^2$. Then for small perturbations $V^2 = u_0^2 + 2u_0u$ and pressure on $2y = H(x, t)$ becomes: $p = p_0(1 - \gamma(\phi_t + \frac{1}{2}2u_0u)/c_0^2 + \dots)$ from where $p_0 - p = \rho_0(\phi_t + u_0\phi_x)$.

2. Boundary condition on the surface $2y = H(x, t)$ must be imposed for potential obtained from velocity in x direction: $\phi_y = H_t + u_0 H_x$.

For function H we shall consider initial and boundary conditions:

$H(x, 0) = H_0 f(x)$, and $H(0, t) = H(L, t) = 0$, $H_{xx}(0, t) = H_{xx}(L, t) = 0$.

Looking for motion **through simple waves** we consider:

$\phi(x, y, t) = F(k_1 x + k_2 y - \omega t)$, $H(x, t) = h(k_1 x - \omega t)$, $F \in C^{(2)}(\mathbf{R})$, $h \in C^{(4)}(\mathbf{R})$.

From (2.1) we find:

$$(k_1^2 + k_2^2 - (\omega/c_0 - M_0 k_1)^2)F'' = 0, \\ \omega^2 h'' + Dk_1^4 h^{(4)} = 2\rho_0(k_1 u_0 - \omega)F'. \quad (2.2)$$

For $F'' \neq 0$ one find the **dispersion equation**:

$$k_2 = \pm \sqrt{\left(\frac{\omega}{c_0} - M_0 k_1\right)^2 - k_1^2} = \pm \sqrt{\frac{(\omega - k_1 u_0)^2}{c_0^2} - k_1^2}.$$

In order to solve equation (2.1)₁ we change the variables (x, t) through (ξ, η) with: $\xi = x - (u_0 + c_0)t$, $\eta = x - (u_0 - c_0)t$ obtaining a new equation:

$$4\phi_{\xi\eta} + \phi_{yy} = 0, \quad \phi = \phi(\xi, \eta, y). \quad (2.3)$$

that will be solved considering a separation of variables (ξ, η) from y variables:

$\phi = \phi(\xi, \eta)F(y)$. We find $\frac{\phi_{\xi\eta}}{\phi} = -\frac{1}{4} \frac{F''(y)}{F(y)} = K$, $K > 0$ for bounded solutions on y.

Then $F(y) = A \sin 2\lambda y + B \cos 2\lambda y$ and $\phi_{\xi\eta} - \lambda^2 \phi = 0$. Solution obtained for $\lambda = 0$ is $\phi(\xi, \eta) = C\xi + D\eta$ and for $\lambda \neq 0$ is $\phi(\xi, \eta) = Ce^{\lambda(\xi+\eta)} + De^{-\lambda(\xi+\eta)}$. We can write the solution of (2.1)₁:

$$\phi(x, y, t) = [Ce^{2\lambda(x-u_0t)} + De^{-2\lambda(x-u_0t)}][A \sin 2\lambda y + B \cos 2\lambda y]. \quad (2.4)$$

We remark that for solution (2.4) of (2.1)₁ we have $\phi_t + u_0\phi_x = 0$.

Solving equation (2.1)₂ we obtain for function $H(x, t)$ the problem:

$$\begin{aligned} H_{tt} + DH_{xxxx} &= 0, \quad D = [0, L] \times (0, \infty), \\ H(x, 0) &= H_0 f(x), \\ H(0, t) &= H(L, t) = 0, \quad H_{xx}(0, t) = H_{xx}(L, t) = 0. \end{aligned} \quad (2.5)$$

Considering a separation of variables $H(x, t) = X(x)T(t) \Rightarrow \frac{T''}{T} = -D \frac{X^{(4)}}{X} = \alpha$

and for $\alpha > 0$ we can write $T'' - \alpha T = 0$, $X^{(4)} + (\alpha/D)X = 0$.

With $\beta = \sqrt[4]{\alpha/D}/\sqrt{2}$, general solution for (2.5) is:

$$\begin{aligned} X(x) &= \cos \beta x \cdot (C_1 e^{\beta x} + C_3 e^{-\beta x}) + \sin \beta x \cdot (C_2 e^{\beta x} + C_4 e^{-\beta x}), \\ T(t) &= Pe^{\sqrt{\alpha}t} + Qe^{-\sqrt{\alpha}t}, \quad P = 0, \text{ for a bounded solution on t.} \end{aligned} \quad (2.6)$$

Solution for equation (2.5)₁ is:

$$H(x, t) = \{\cos \beta x [C_1 e^{\beta x} + C_3 e^{-\beta x}] + \sin \beta x [C_2 e^{\beta x} - C_4 e^{-\beta x}]\} Q e^{-\sqrt{\alpha}t} \quad (2.7)$$

From conditions (2.5)₂ we find:

$$\begin{aligned} H(0, t) = 0 &\Rightarrow C_3 = -C_1, \quad H_{xx}(0, t) = 0 \Rightarrow C_2 = C_4, \\ H(L, t) = 0 &\Rightarrow C_1 \cos \beta L (e^{\beta L} - e^{-\beta L}) + \sin \beta L (C_2 e^{\beta L} + C_4 e^{-\beta L}) = 0, \\ H_{xx}(L, t) = 0 &\Rightarrow 2\beta L = -\frac{\pi}{2} + 2k\pi, \quad k \in N \Rightarrow \sqrt{\alpha} = \sqrt{2D}(-\pi/4 + k\pi)^2/L^2, \end{aligned} \quad (2.8)$$

and (2.7) $\Rightarrow C_2 = C_1 t h(-\pi/4 + k\pi)$ for each k .

The general solution for (2.5) is:

$$H(x,t) = \sum_{k=1}^{\infty} M_k \{ \cos((-\pi/4 + k\pi)x/L) sh((-\pi/4 + k\pi)x/L) + \\ + t h(-\pi/4 + k\pi) \sin((-\pi/4 + k\pi)x/L) ch((-\pi/4 + k\pi)x/L) \} e^{-\frac{\sqrt{2D}}{L^2} (-\frac{\pi}{4} + k\pi)^2 t} \quad (2.9)$$

with M_k determined from $H(x,0) = H_0 f(x)$ for which:

$$H_0 f(x) = \sum_{k=1}^{\infty} M_k \{ \cos((-\pi/4 + k\pi)x/L) sh((-\pi/4 + k\pi)x/L) + \\ + t h(-\pi/4 + k\pi) \sin((-\pi/4 + k\pi)x/L) ch((-\pi/4 + k\pi)x/L) \}. \quad (2.10)$$

Conclusions

For one-dimensional case was investigated the influence of Mach number of the unperturbed flow upon the speed of propagating waves (presented in Figs 2 and 3).

For the bi-dimensional case instead of a discussion of the dispersion equation we have studied the solution for potential using condition on the boundary: $\phi_y = H_t + u_0 H_x$ on $2y = H(x,t)$, and initial conditions on velocity. The velocity field is expressed by:

$$u = \phi_x = 2\lambda [C e^{2\lambda(x-u_0 t)} - D e^{-2\lambda(x-u_0 t)}] [A \sin 2\lambda y + B \cos 2\lambda y], \\ v = \phi_y = 2\lambda [C e^{2\lambda(x-u_0 t)} + D e^{-2\lambda(x-u_0 t)}] [A \cos 2\lambda y - B \sin 2\lambda y].$$

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